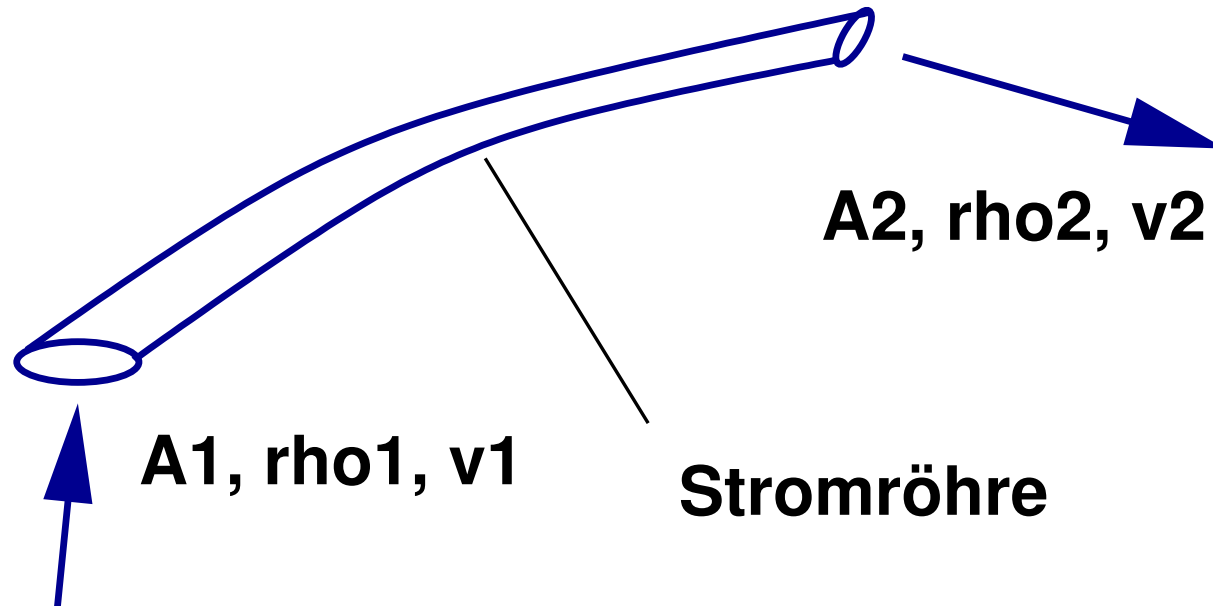


Hydrodynamics

Continuity equation



Conservation of mass:

$$\underbrace{\rho_1 v_1 A_1}_{\dot{m}_1} = \underbrace{\rho_2 v_2 A_2}_{\dot{m}_2} \quad \text{Mass flux}$$

inkompressible fluid: ($\rho_1 = \rho_2 = \text{const}$)

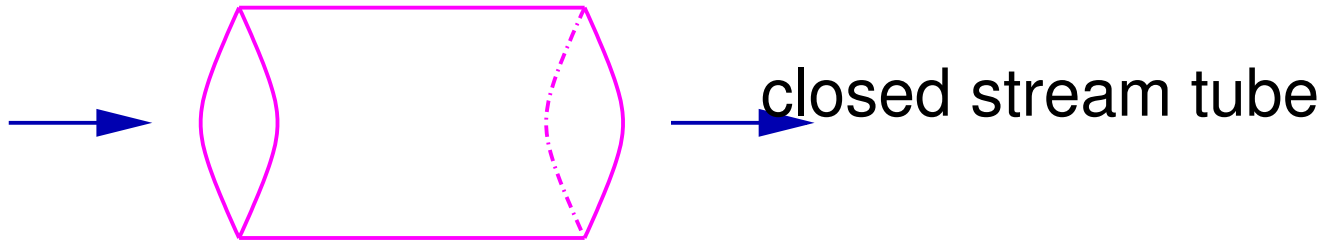
Conservation of volume flux :

$$\underbrace{v_1 A_1}_{\dot{Q}_1} = \underbrace{v_2 A_2}_{\dot{Q}_2} \quad \text{Volume flux}$$

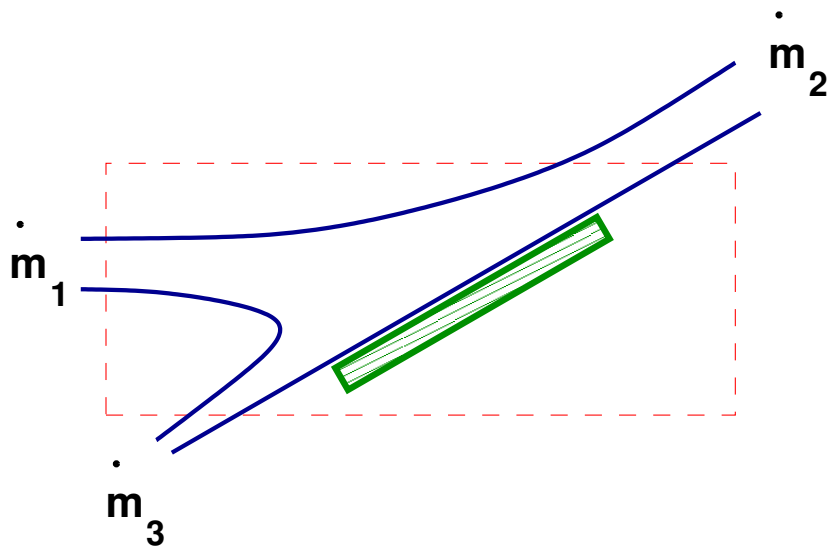
Hydrodynamics

Example

pipe flow: $A = \text{const}$



water jet

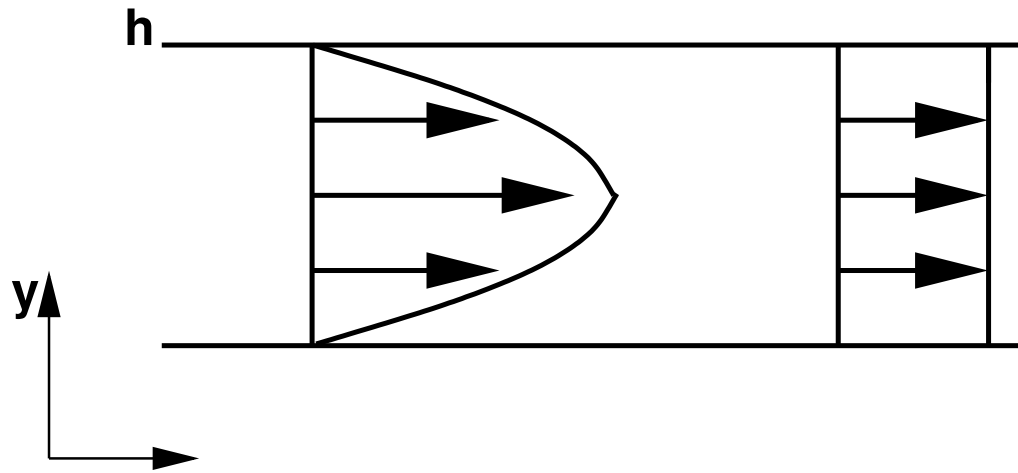


closed control volume

$$\dot{m}_1 = \dot{m}_2 + \dot{m}_3$$

Continuity

Important: In the 1-dimensional continuity equation \bar{v} is an average value of the velocity. In reality \bar{v} is not constant due to friction, vortices, ...!



x reality
 $\bar{v} = \bar{v}(y)$

\bar{v} ist constant
in the continuity equation

mass flux must be constant

$$\longrightarrow \int \rho v(y) dy = \rho \bar{v} h$$

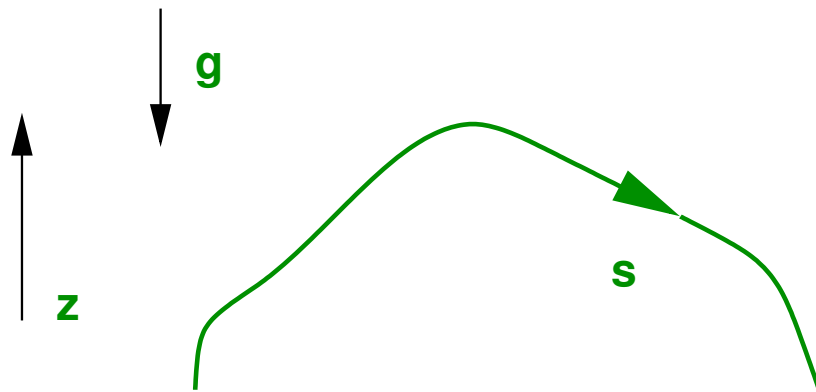
Bernoulli

2. Newtonian law:

mass × acceleration = sum of outer forces

$$m \cdot \frac{d\vec{v}}{dt} = \Sigma F_a$$

Equation of motion for an infinitesimal element along one streamline



$$\underbrace{\rho \frac{d\vec{v}}{dt}}_{\text{inertia}} = \underbrace{-\frac{\partial p}{\partial s}}_{\text{pressure}} - \underbrace{\rho g \frac{dz}{ds}}_{\text{gravitation}} - \underbrace{R'}_{\text{friction}}$$

Bernoulli

along a streamline: $v = v(s, t)$

$$d\vec{v} = \frac{\partial \vec{v}}{\partial t} dt + \frac{\partial \vec{v}}{\partial s} ds$$

$$\longrightarrow \frac{d\vec{v}}{dt} = \frac{\partial \vec{v}}{\partial t} + \frac{ds}{dt} \frac{\partial \vec{v}}{\partial s} = \frac{\partial \vec{v}}{\partial t} + v \frac{\partial \vec{v}}{\partial s}$$

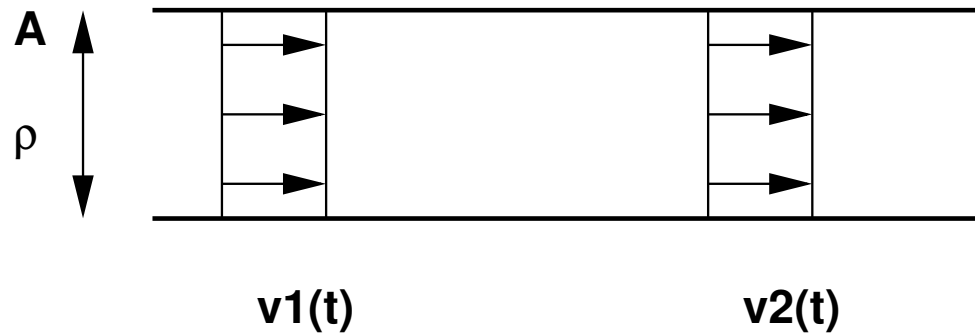
total
(substantial)
acceleration
of a particle

lokal acceleration

convective acceleration

example

Rohrströmung

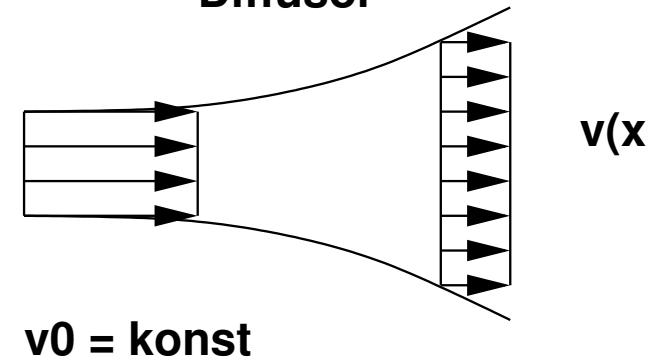


$$A, \rho = konst$$

$$\longrightarrow v_1(t) = v_2(t)$$

only local acceleration

Diffusor



only convective acceleration

example

assumptions:

- incompressible ($\rho = const$)
- frictionless ($R' = 0$)
- steady $\frac{\partial}{\partial t} = 0$
- constant gravitation ($\vec{g} = const$)

$$\rho \left[\frac{\partial v}{\partial t} + v \frac{\partial v}{\partial s} \right] = - \frac{\partial p}{\partial s} - \rho g \frac{dz}{ds} - R'$$

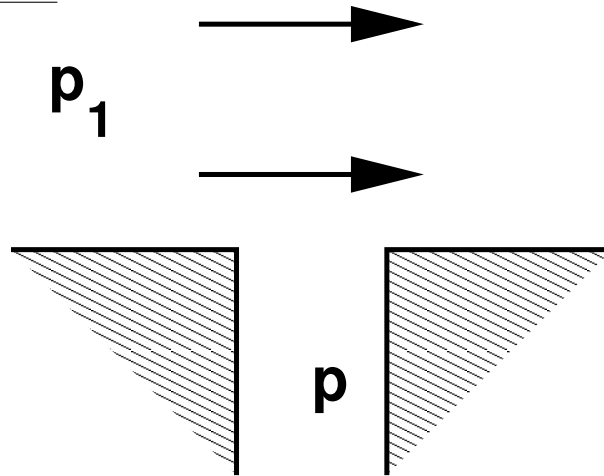
$$\quad \quad \quad = 0 \quad \quad \quad \quad \quad \quad \quad = 0$$

$$f(s) \longrightarrow \frac{\partial}{\partial s} = \frac{d}{ds}$$

$$\frac{1}{2} \rho \frac{dv^2}{ds} = - \frac{dp}{ds} - \rho g \frac{dz}{ds} \longrightarrow \frac{\rho}{2} v^2 + p + \rho g z = const$$

pressure measurement

static pressure: p (Index: 1, 2, a, ∞)

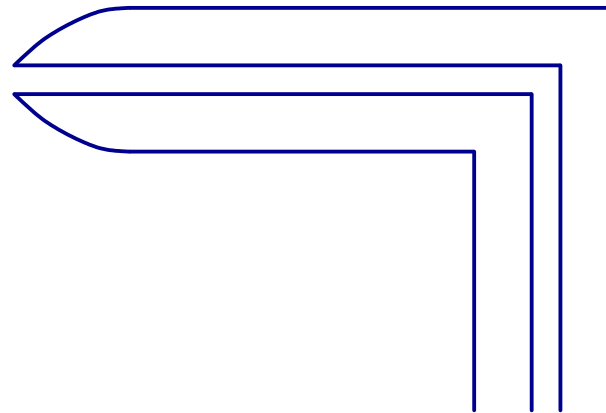


Total pressure (pitot tube): p_0, p_{01}, p_{02}, p_t

$$p_0 = p + \frac{1}{2}\rho v^2 + \rho g h$$

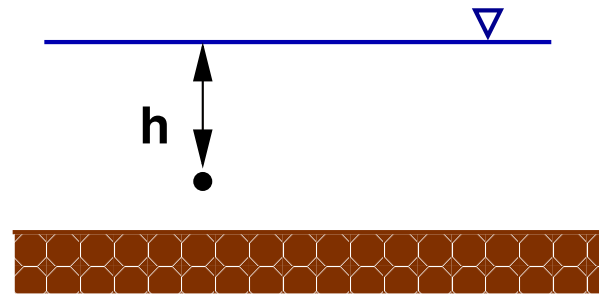
at constant height $\Delta h = 0$

$$\longrightarrow p_0 = p + \frac{1}{2}\rho v^2$$



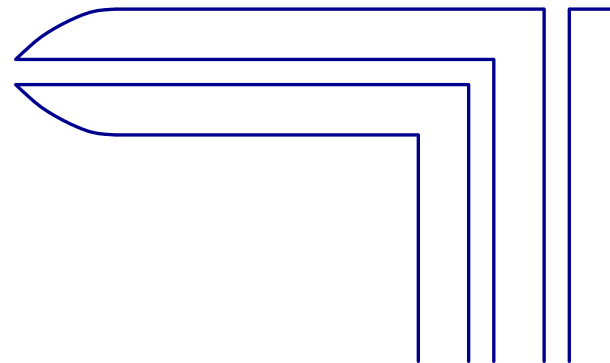
pressure measurement

potential pressure: $p_{pot} = \rho g h$



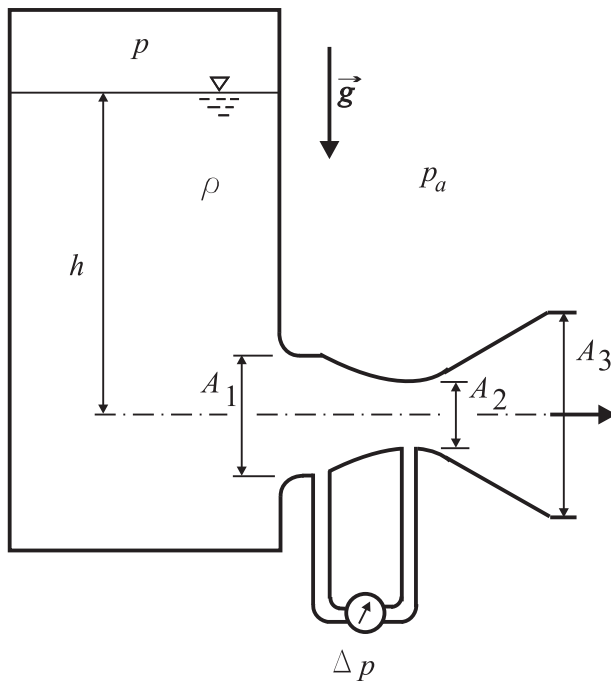
dynamical pressure: $p_{dyn} = \frac{1}{2}\rho v^2$

kinetic energy is converted, when the flow is decelerated to $\vec{v} = 0$



6.4

Water flows from a large pressurized tank into the open air. The pressure difference Δp is measured between A_1 and A_2 .



$$\begin{aligned}
 A_1 &= 0,3 \text{ m}^2, & A_2 &= 0,1 \text{ m}^2, \\
 A_3 &= 0,2 \text{ m}^2, & h &= 1 \text{ m}, \\
 \rho &= 10^3 \text{ kg/m}^3, & p_a &= 10^5 \text{ N/m}^2, \\
 \Delta p &= 0,64 \cdot 10^5 \text{ N/m}^2 & g &= 10 \text{ m/s}^2
 \end{aligned}$$

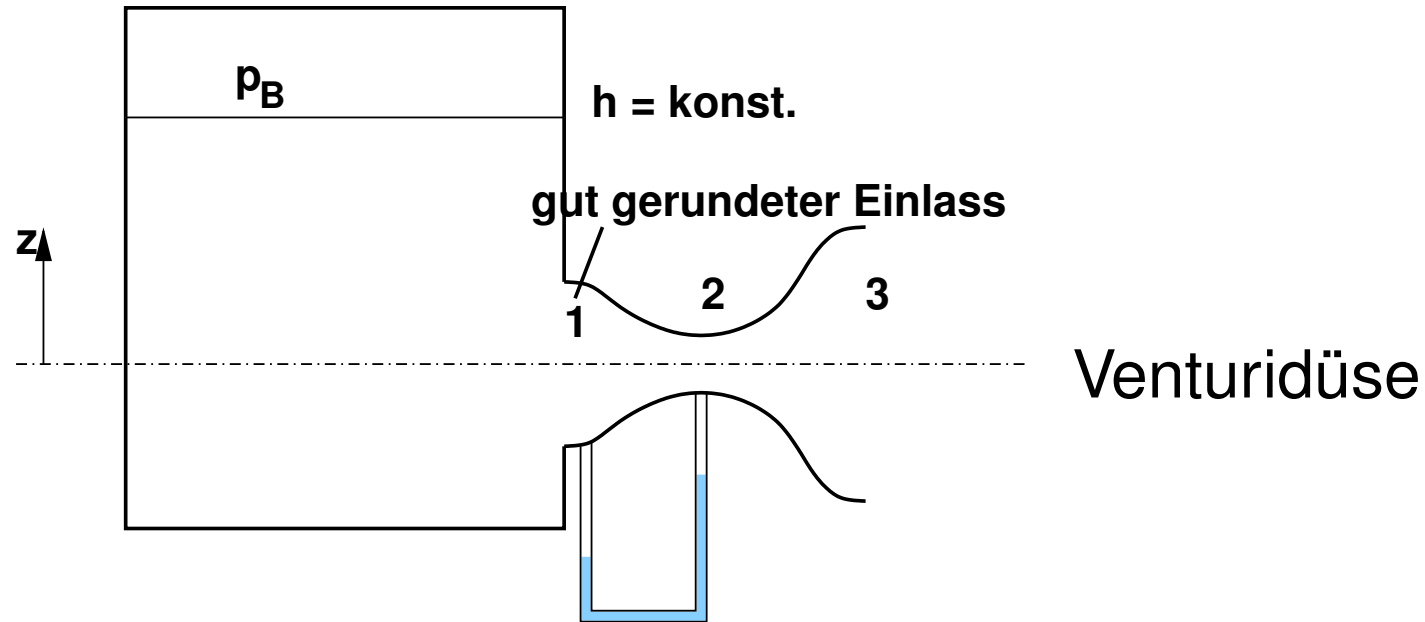
Compute the

a) velocities v_1 , v_2 , v_3 ,

b) pressures p_1 , p_2 , p_3 and the pressure p above the water surface!

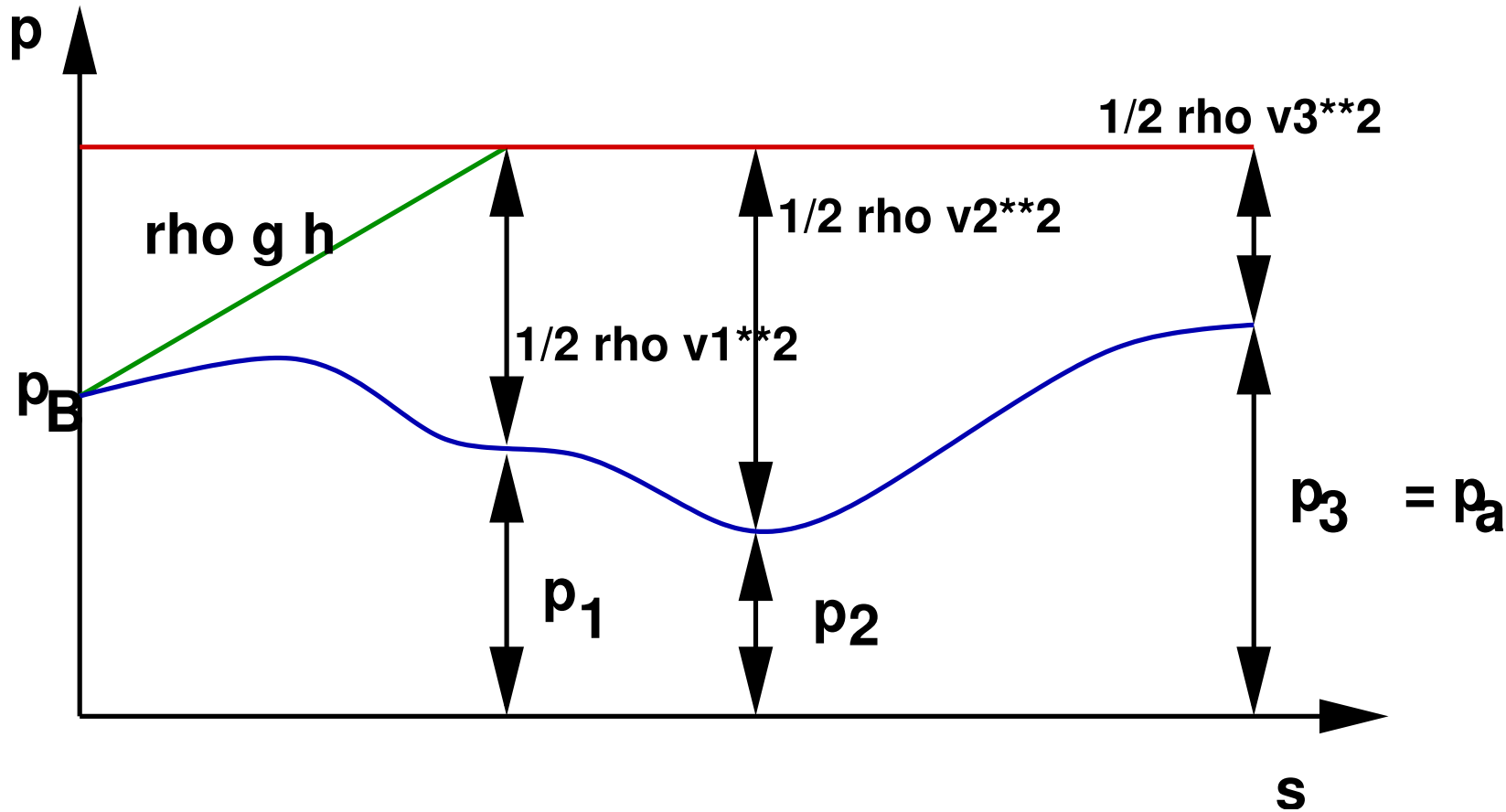
6.4

pressure tank with nozzle



6.4

conservation of total energy along a streamline → qualitatively



$$\text{Bernoulli: } p_0 = p_b + \rho g h = p_i + \frac{1}{2} \rho v_i^2$$

continuity (mass balance): $\implies \dot{m} = \rho \dot{Q} = \text{const.}$

$$\rho = \text{const} \implies v_1 A_1 = v_2 A_2 = v_3 A_3 \implies A \downarrow \implies v \uparrow \implies p \downarrow$$

a) measured $\Delta p = p_1 - p_2$ Bernoulli: $p_1 + \frac{\rho}{2} v_1^2 = p_2 + \frac{\rho}{2} v_2^2$

$$\implies \Delta p = p_1 - p_2 = \frac{\rho}{2} (v_2^2 - v_1^2) > 0$$

$$v_1 = v_2 \frac{A_2}{A_1} \rightarrow \Delta p = \frac{\rho}{2} \left[1 - \frac{A_2^2}{A_1^2} \right] v_2^2 \longrightarrow v_2 = \sqrt{\frac{2 \Delta p}{\rho \left(1 - \left(\frac{A_2}{A_1} \right)^2 \right)}} = 12 \frac{m}{s}$$

$$v_1 = v_2 \frac{A_2}{A_1} = 4 \frac{m}{s} \qquad v_3 = v_2 \frac{A_2}{A_3} = 6 \frac{m}{s}$$

6.4

The Venturi-nozzle is used to measure mass- and volume fluxes!

$$\dot{Q} = vA = v_2A_2$$

prinziple:

- measurement of Δp
- computation of v_2
- computation of volume- and massflux

6.4

b) determination of pressures p_B, p_1, \dots, p_3

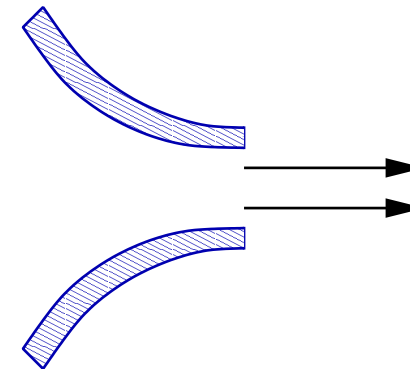
p_0 represents the energy that can be converted into kinetic energy

$$p_0 = p_B + \rho gh = p_1 + \frac{\rho}{2}v_1^2 = p_2 + \frac{\rho}{2}v_2^2 = p_3 + \frac{\rho}{2}v_3^2$$

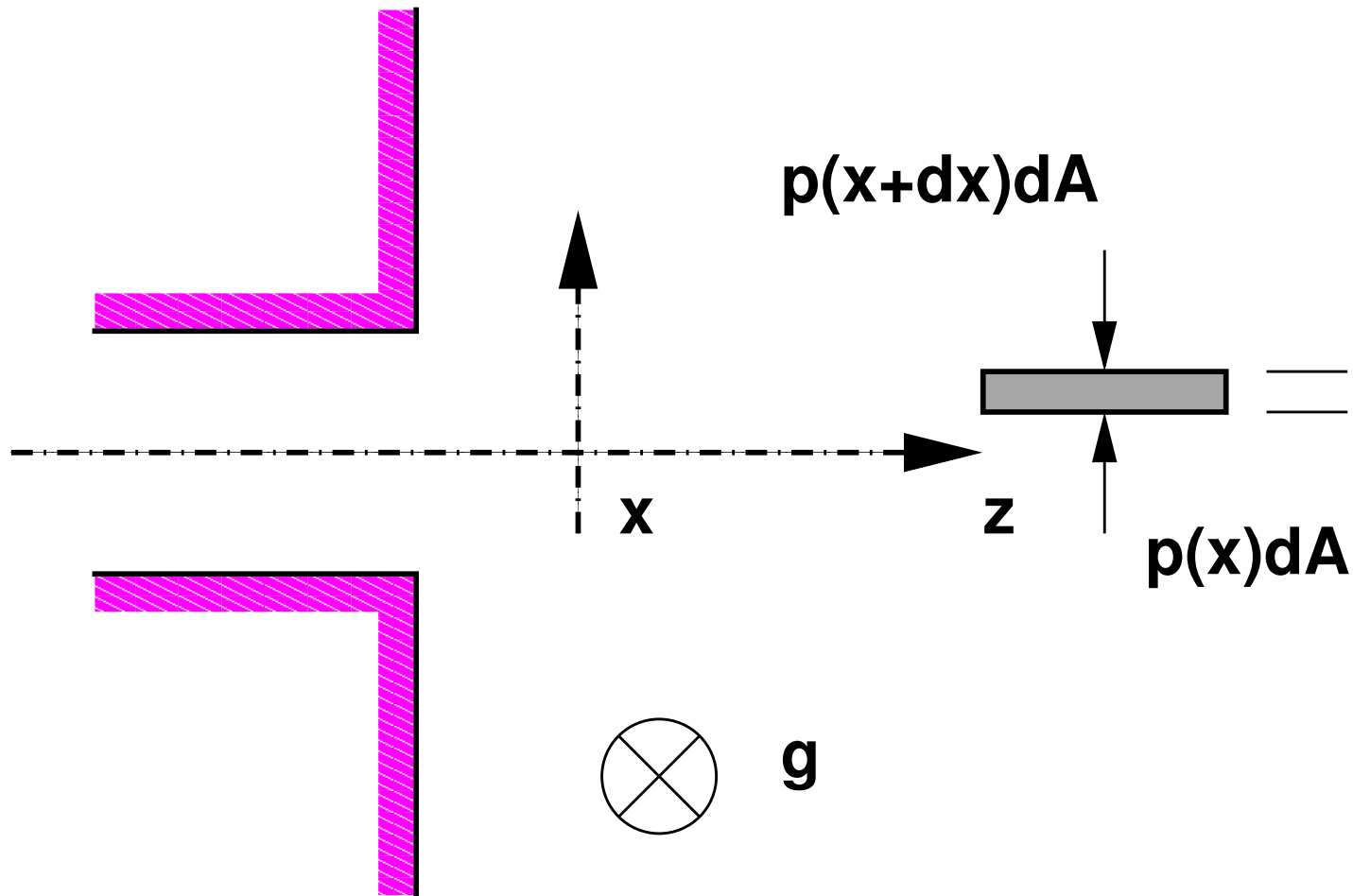
If we know one pressure, we can compute the other values by using Bernoulli's equation

p_3 in the exit cross section

Assumption: parallel streamlines at the sharp edged exit



equation of motion for an element



equation of motion in x -direction for a moving control volume $dA dx$ (includes always the same particles!)

equation of motion for an element

$$m \frac{du}{dt} = \ddot{x} \rho dA dx = p(x) dA - p(x + dx) dA$$

$$\longrightarrow \ddot{x} \rho dA dx = p(x) dA - \left(p + \frac{\partial p}{\partial x} dx \right) dA$$

$$\longrightarrow \rho \ddot{x} = - \frac{\partial p}{\partial x}$$

Assumption: parallel stream lines

$$\longrightarrow \dot{x} = 0 \quad \text{velocity} \quad u = \frac{dx}{dt} = \dot{x}$$

$$\longrightarrow \quad \text{necessary condition: } \ddot{x} = 0 \longrightarrow \frac{\partial p}{\partial x} = 0$$

\implies the pressure in the exit cross-section is function of y

flow into air: $\frac{dp}{dy} = -\rho g$

Neglect the potential energy \longrightarrow $P_{exit} = P_{ambience} = const.$

6.4

$$p_3 = p_a$$

Remark:

Bernoulli: $0 \rightarrow 3$

$$p_B + \rho gh = p_a + \frac{1}{2}\rho v_3^2$$

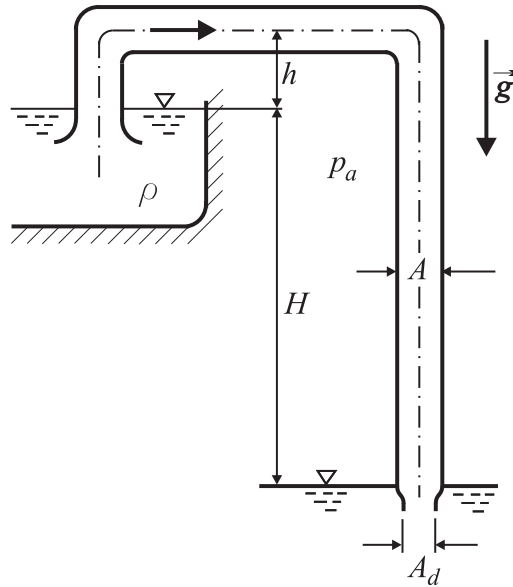
$$\rightarrow v_3 = \sqrt{\frac{2}{\rho} (p_B - p_a + \rho gh)}$$

open tank $p_B = p_a$

$\rightarrow v_3 = \sqrt{2gh} \neq f(A_3)$ theorem of Torricelli (15.Okt. 1608 - 25.Okt. 1647)

6.5

Two large basins located one upon the other are connected with a duct.



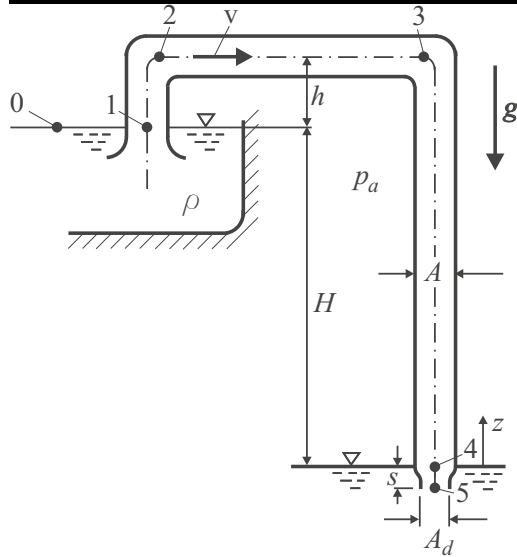
$$A = 1 \text{ m}^2, \quad A_d = 0,1 \text{ m}^2, \quad h = 5 \text{ m}, \quad H = 80 \text{ m},$$

$$p_a = 10^5 \text{ N/m}^2, \quad \rho = 10^3 \text{ kg/m}^3, \quad g = 10 \text{ m/s}^2$$

- Determine the volume rate!
- Outline the distribution of static pressure in the duct!
- At what exit cross section bubbles are produced, when the vapour pressure is

$$p_D = 0,025 \cdot 10^5 \text{ N/m}^2?$$

6.5



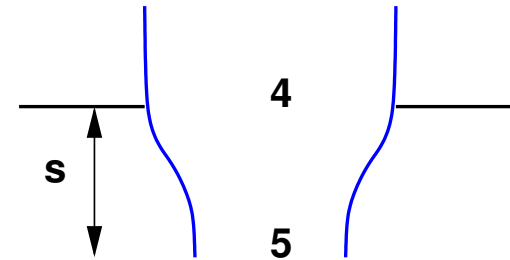
incompressible, frictionless, steady

→ Bernoulli

a) volume flux: $\dot{Q} = vA = v_5A_5$

Bernoulli: 0 → 5

$$p_a + \rho gH = p_5 + \rho g(-s) + \frac{\rho}{2}v_5^2$$



computation of p_5 :

straight parallel streamlines

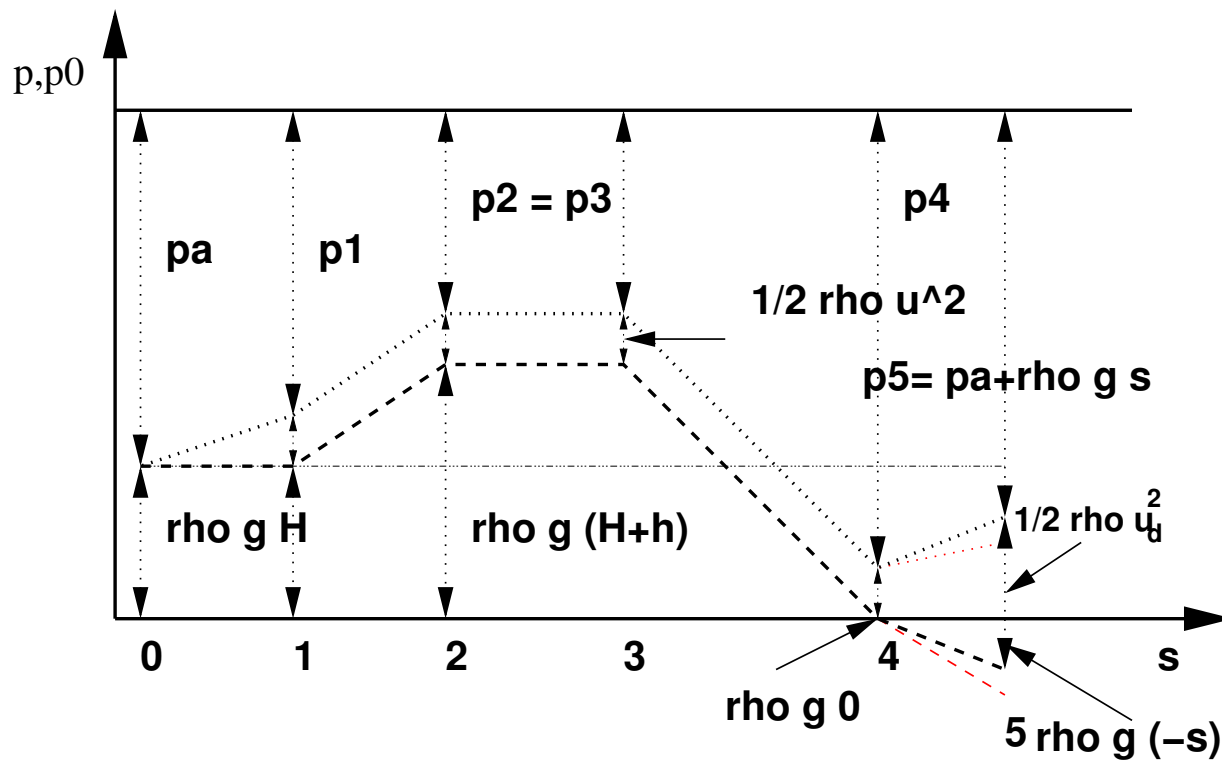
→ pressure is constant
in the exit cross-section

$$\rightarrow p_5 = p_a + \rho g s$$

$$p_a + \rho g H = p_a + \rho g s - \rho g s + \frac{1}{2} \rho v_5^2 \longrightarrow v_5 = \sqrt{2gH} \neq f(A_d, s)$$

$$\longrightarrow \dot{Q} = A_d v_5 = 4 \frac{m^3}{s}$$

b)



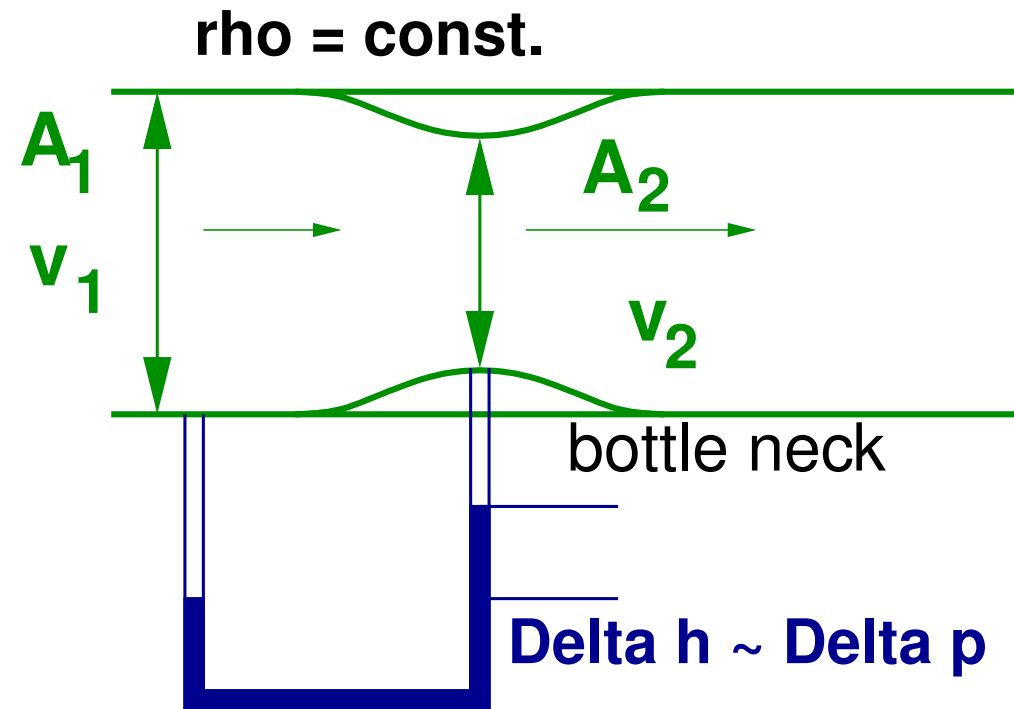
6.5

c) minimum pressure between 2 and 3: $p_2 = p_3 = p_D$

continuity: $v_5^* A_D^* = v^* A$

Bernoulli: $p_a = p_D + \rho g h + \frac{1}{2} \rho v^{*2}$

$$\longrightarrow A_d = A \sqrt{\frac{p_a - p_D}{\rho g H} - \frac{h}{H}} = 0.244 \text{ m}^2$$



Theoretical volume flux: \dot{Q}_{th} for frictionless flow

1. Bernoulli: $p_1 + \frac{\rho}{2}v_1^2 = p_2 + \frac{\rho}{2}v_2^2$
2. continuity: $v_1A_1 = v_2A_2$

extended Bernoulli

ratio of areas: $m = \frac{A_2}{A_1}$: \rightarrow conti $v_1 = v_2 m$

$$\rightarrow \text{Bernoulli: } \frac{p_1}{\rho} + \frac{1}{2} v_2^2 m^2 = \frac{p_2}{\rho} + \frac{1}{2} v_2^2$$

$$\rightarrow v_2^2 (1 - m^2) = 2 \frac{p_1 - p_2}{\rho} = 2 \frac{\Delta p}{\rho}$$

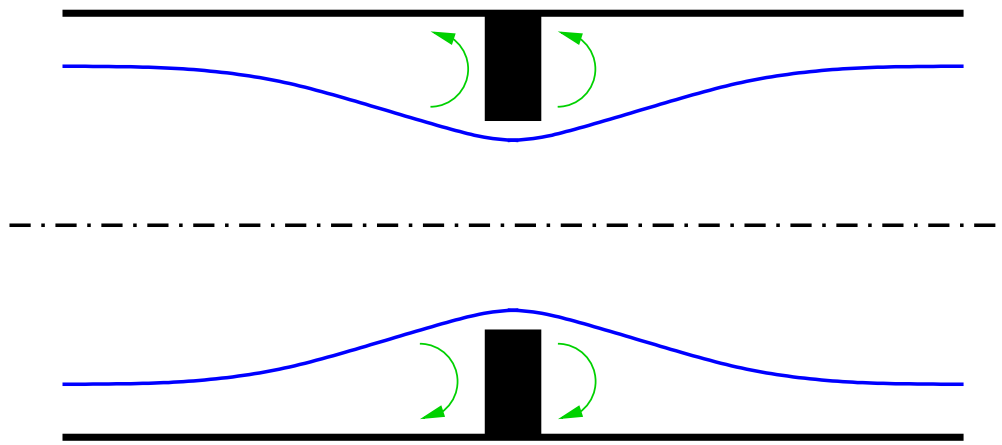
$$\rightarrow v_2 = \sqrt{\frac{2\Delta p}{\rho(1 - m^2)}}$$

$$\rightarrow \dot{Q}_{th} = A_2 \sqrt{\frac{2\Delta p}{\rho(1 - m^2)}}$$

extended Bernoulli (Cont'd)

In reality losses from friction, vortices, bottle necks ... occur.
 → the flow is no longer frictionless

The losses and the ration of aread are put together in the
 discharge coefficient α



vortex, dissipation

$$\dot{Q}_{real} = \alpha A_2 \sqrt{\frac{2\Delta p}{\rho(1 - m^2)}}$$

$$\alpha^* = \alpha \sqrt{\frac{1}{1 - m^2}}$$

α^* from experiments

losses in pipe flows can be predicted similar.

extended Bernoulli (Cont'd)

pressure loss across a constructive element (elbow, valve, ...)

$$\Delta p_v = \zeta \cdot \frac{1}{2} \rho v^2$$

coefficient: $\zeta = \frac{\Delta p_v}{\frac{1}{2} \rho v^2} = \frac{\text{pressure loss}}{\text{dynamic pressure}}$

$$\longrightarrow v = \frac{1}{\sqrt{\zeta}} \sqrt{\frac{2\Delta p}{\rho(1 - m^2)}} \implies \dot{Q} = v \cdot A = \frac{1}{\sqrt{\zeta}} A \sqrt{\frac{2\Delta p}{\rho(1 - m^2)}}$$

(Experiments, standards \longrightarrow catalogue)

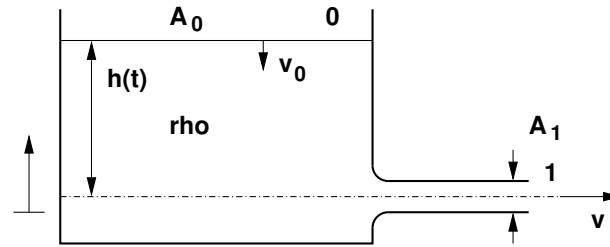
Hydrodynamics: unsteady Bernoulli

assumption

$$\frac{A_1}{A_0} \ll 1 \longrightarrow v_0 \ll v_1$$

v_0 is neglectable

but $h = h(t)$ and $v_1 = v_1(t)$



unsteady Bernoulli from "0" to "1"

$$p_a + \frac{\rho}{2}v_0^2(t) + \rho gh(t) = p_a + \frac{\rho}{2}v_1^2(t) + \int_0^1 \rho \frac{\partial v}{\partial t} ds$$

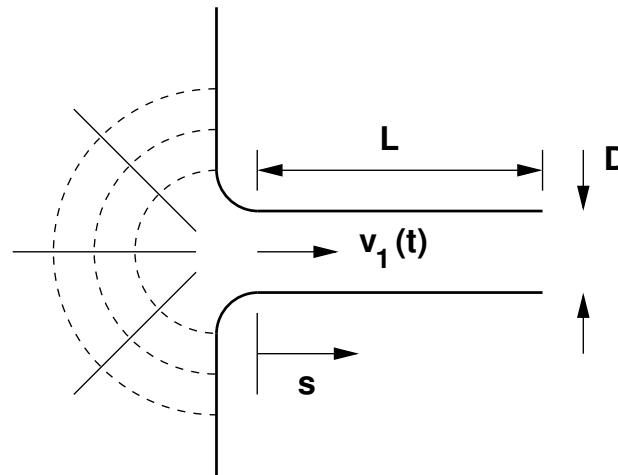
Assumption $v_1(t) = \sqrt{2gh(t)}$

continuity equation: $v_1(t)A_1 = -\frac{dh}{dt}A_0$

→ differential equation for für $h(t)$

Hydrodynamics: unsteady Bernoulli

“well rounded” inlet



assumption:

$$s < -\frac{D}{\sqrt{8}} : \text{radial flow with } \dot{Q} = v \cdot \frac{\pi s^2}{2}$$

$$s \geq -\frac{D}{\sqrt{8}} : v = v_1$$

Hydrodynamics: unsteady Bernoulli

Potential flow without any losses \longrightarrow Bernoulligleichung

$$\begin{aligned} \int_{-\infty}^L \frac{\partial v}{\partial t} ds &= \int_{-\infty}^{-D/\sqrt{8}} \frac{\partial v(s)}{\partial t} ds + \int_{-D/\sqrt{8}}^L \frac{\partial v_1}{\partial t} ds = \int_{-\infty}^{-D/\sqrt{8}} \frac{\partial}{\partial t} \left(\frac{v_1 \pi \frac{D^2}{4}}{2\pi s^2} \right) ds + \int_{-D/\sqrt{8}}^L \frac{\partial v_1}{\partial t} ds \\ &= \frac{dv_1(t)}{dt} \int_{-\infty}^{-D/\sqrt{8}} \frac{D^2}{8s^2} ds + \frac{dv_1(t)}{dt} \int_{-D/\sqrt{8}}^L ds \\ &= \left(\frac{D}{\sqrt{8}} + L + \frac{D}{\sqrt{8}} \right) \cdot \frac{dv_1(t)}{dt} = \underbrace{\left(\frac{D}{\sqrt{2}} + L \right)}_{\text{well runde inlet}} \frac{dv_1(t)}{dt} \end{aligned}$$

if $L \gg D \longrightarrow \int_{-\infty}^L \frac{\partial v}{\partial t} ds = L \cdot \frac{dv_1(t)}{dt}$

Example: duct from a big tank

a) Bernoulli: $p_a + \rho g(h + s) = p_a + \rho g s + \frac{\rho}{2} v_4^2 + \rho \int_{s_0}^{s_4} \frac{\partial v}{\partial t} ds$

well rounded inlet: $\int_{s_0}^{s_4} \frac{\partial v}{\partial t} ds = L \frac{dv_4}{dt}$

$$\longrightarrow \rho g h = \frac{\rho}{2} v_4^2 + L \rho \frac{dv_4}{dt} \quad \longrightarrow \int_0^T dt = L \frac{dv_4}{gh - \frac{v_4^2}{2}}$$

Integration

$$\begin{aligned} T &= 2L \int_0^{0.99\sqrt{2gh}} \frac{dv_4}{2gh - v_4^2} \\ &= \frac{L}{\sqrt{2gh}} \ln \frac{\sqrt{2gh} + v_4}{\sqrt{2gh} - v_4} \Big|_0^{0.99\sqrt{2gh}} = 10.6 \text{ s} \end{aligned}$$

Example (cont'd)

the accelerated initial flow is **dependent** of L ,
but $v_4(t \rightarrow \infty)$ is **independent** of L .

b)

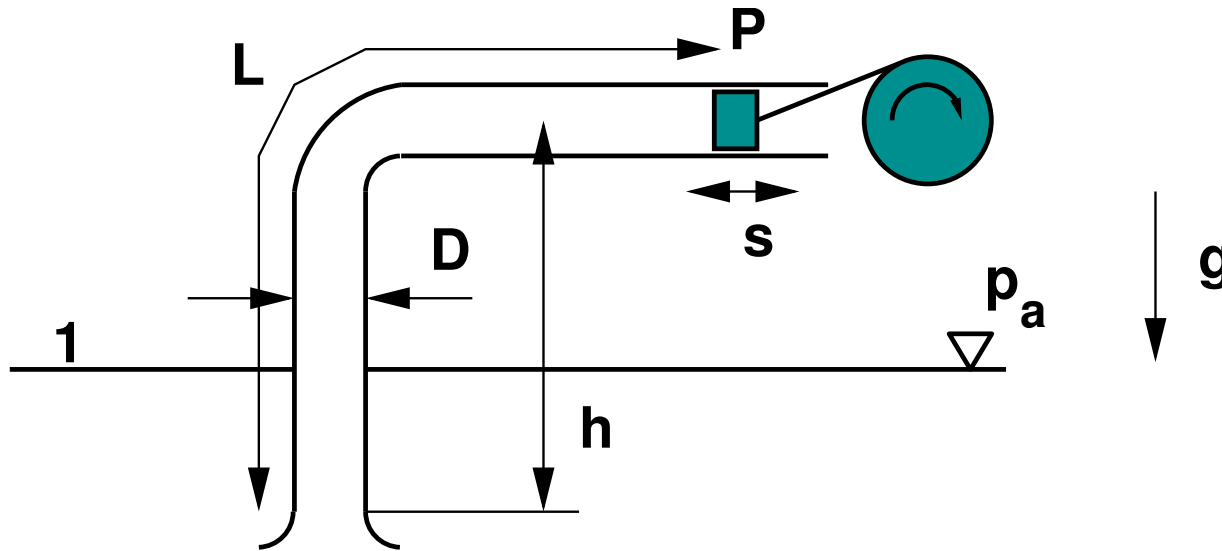
$$p_a = p_A + \rho g h_1 + \frac{\rho}{2} v_4^2 + L_1 \frac{dv_4}{dt}$$

$$t \rightarrow \infty : p_a = p_{A,\infty} + \rho g h_1 + \frac{\rho}{2} 2gh$$

$$\text{from a) } \frac{dv_4}{dt} = \frac{1}{L} \left(gh - \frac{v_4^2}{2} \right)$$

$$\implies p_A - p_{A,\infty} = \rho g h \left(1 - 0.99^2 \right) \left(1 - \frac{L_1}{L} \right) = 746 \frac{N}{m^2}$$

example: moving piston



A piston is moving in a duct: $s = s_0 \cdot \sin \omega t$

$$p_a = 1 \text{ bar} \quad L = 10 \text{ m} \ll D \quad h = 2 \text{ m} \quad g = 10 \frac{\text{m}}{\text{s}^2}$$

$$s_0 = 0.1 \text{ m} \quad \rho = 10^3 \frac{\text{kg}}{\text{m}^3} \quad p_D = 2500 \frac{\text{N}}{\text{m}^2}$$

At what angular speed ω the pressure at the bottom of the piston reaches the vapour pressure p_D ?

example: moving piston

$$p_a = p_P + \rho gh + \frac{\rho}{2} v_P^2 + \rho \int_{s_1}^{s_P} \frac{\partial v}{\partial t} ds$$

$$s_0 \ll L \longrightarrow \int_{s_1}^{s_P} \frac{\partial v}{\partial t} ds = L \frac{dv_P}{dt}$$

$$p_P = p_a - \rho gh + \rho s_0 \omega^2 \left(L \sin \omega t - \frac{s_0}{2} \cos^2 \omega t \right)$$

$$p_{P,\min} = p_D$$

$$p_D = p_{P,\min} \text{ at } \cos \omega t = 0 \longrightarrow \frac{dp_P}{dt} = 0$$

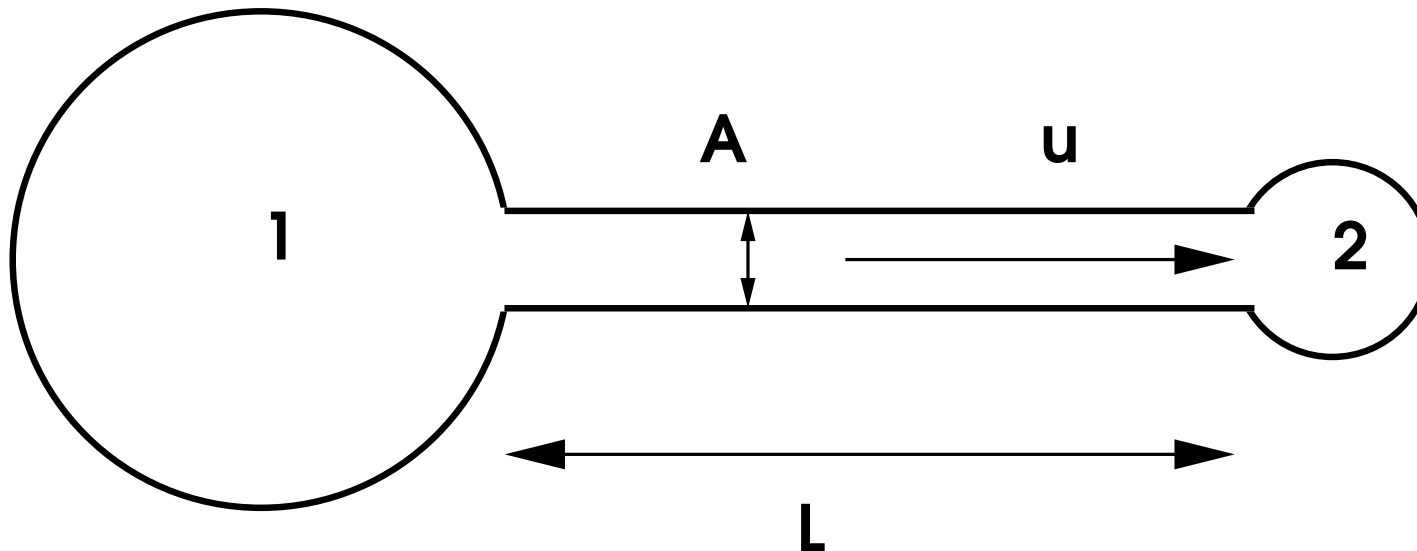
$$\implies \omega = \sqrt{\frac{p_a - p_D - \rho gh}{\rho s_0 L}} = 8.8 \text{ s}^{-1}$$

example: balloons

2 balloons are connected with a pipe of length L and the cross-section A . The pressure in the balloons depends linearly on the balloon volume. V_0 is the volume at ambient pressure.

$$p = p_a + C(V - V_0)$$

At $t = 0$ one balloon is compressed by the volume ΔV .



example: balloons

- a) Show, that the frictionless flow in the pipe is described by the equation of oscillation

$$\ddot{u} + K^2 u = 0.$$

Determine the eigenfrequency of the system.

- b) Determine the maximum of the velocity in the pipe.
c) Determine the maximum of the pressure difference between the balloons.

Given: $\Delta V, L, A, \rho, C$

Hint:

- General Ansatz for the equation of oscillations

$$\ddot{x} + a^2 x = 0:$$

$$x = C_1 \sin(at) + C_2 \cos(at)$$

example: balloons

a)

continuity for the balloons:

$$\dot{V}_1 = -uA \quad \dot{V}_2 = uA$$

Bernoulli (unsteady, frictionless)

$$\rho L \dot{u} + p_2 - p_1 = 0 \quad \left| \left(\frac{\partial}{\partial t} \right) \right.$$

$$\implies \rho L \ddot{u} + \dot{p}_2 - \dot{p}_1 = 0 \quad p = p_a + C(V - V_0) \implies \dot{p} = C\dot{V}$$

$$\implies \rho L \ddot{u} + 2CAu = 0 \implies \ddot{u} + \frac{2CA}{\rho L} u = 0 \implies K = \sqrt{\frac{2CA}{\rho L}}$$

example: balloons

b)

$$V_2(t) = V + 0 - \Delta V \cos(Kt) \Rightarrow \dot{V}_2(t) = \Delta V K \sin(Kt) = uA$$

$$\Rightarrow u(t) = \frac{\Delta V K}{A} \sin(Kt) \Rightarrow u_{max} = \frac{\Delta V K}{A} = \Delta V \sqrt{\frac{2C}{\rho AL}}$$

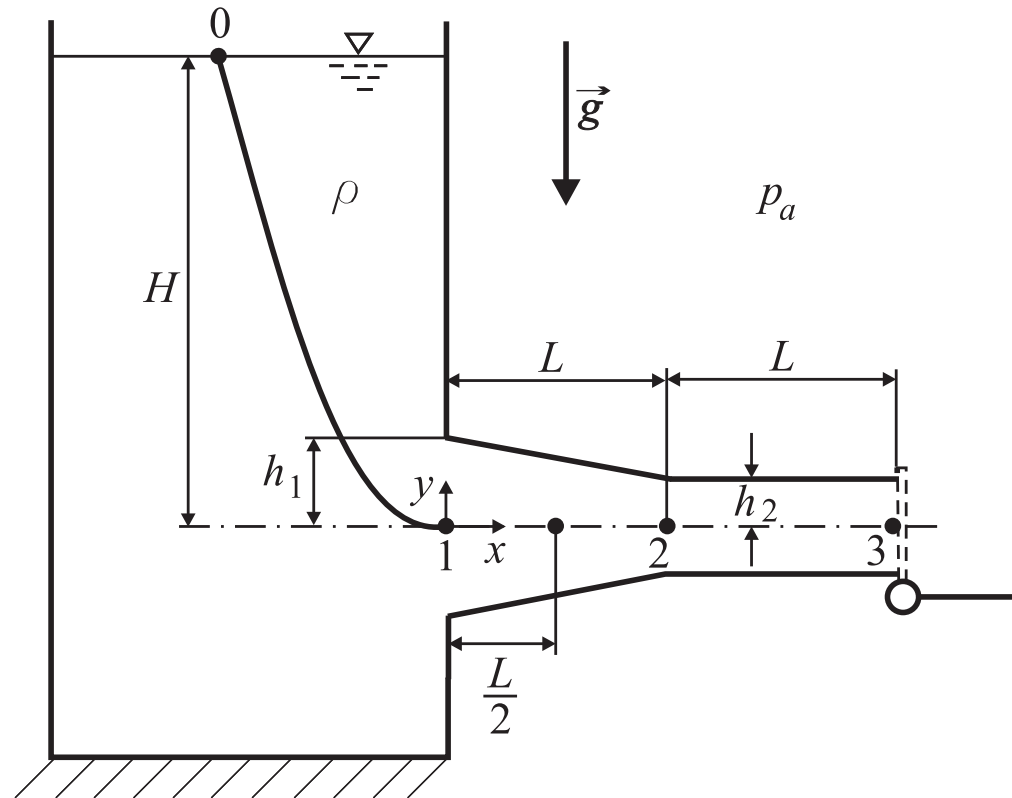
c)

$$|p_2 - p_1|_{max} = |\rho L \dot{u}|_{max}$$

$$\dot{u} = \frac{\Delta V K^2}{A} \cos(Kt) \Rightarrow |p_2 - p_1|_{max} = \frac{\rho L \Delta V}{A} K^2 = 2C \Delta V$$

6.6

The flap at the exit of the water pipe (constant width B) of a large container is opened abruptly. The appearing flow is without any losses.



Given: H, h_1, h_2, g, L ; $L \gg h_1$

6.6

Determine

a) the differential equation for the exit velocity v_3

b) - the local acceleration

- the convective acceleration

- the substantial acceleration

at $x = \frac{L}{2}$ when the exit velocity reaches half of its asymptotic final value!

Hint: The computation of $v(t)$ is not necessary for solving this problem.

6.6

Bernoulli from "0" to "3"

$$p_a + \rho g H = p_3 + \frac{\rho}{2} v_3^2 + \int_0^3 \rho \frac{\partial v}{\partial t} ds, \quad p_3 = p_a$$

Splitting of the integral

$$\int_0^1 \rho \frac{\partial v}{\partial t} ds \approx 0 (h_1 \ll L)$$

$$\int_1^2 \rho \frac{\partial v}{\partial t} ds, \quad v = v_2 \frac{h_2}{h}, \quad h = h_1 + \frac{h_2 - h_1}{L} x$$

$$\Rightarrow \rho \frac{dv_2}{dt} \int_1^2 \frac{h_2}{h_1 + \frac{h_2 - h_1}{L} x} dx = \rho \frac{dv_2}{dt} \frac{h_2 L}{h_2 - h_1} \ln \frac{h_2}{h_1} = \rho \frac{dv_2}{dt} \bar{L}$$

$$\rho \int_2^3 \frac{\partial v}{\partial t} ds = \rho L \frac{dv_2}{dt}$$

introduce in Bernoulli

$$p_a + \rho g H = p_a + \frac{\rho}{2} v_3^2 + \rho \frac{dv_3}{dt} (\bar{L} + L)$$

$$\frac{dv_3}{dt} = \frac{1}{\bar{L} + L} \left(g H - \frac{v_3^2}{2} \right)$$

$$t \rightarrow \infty : \quad g H - \frac{1}{2} v_{3e}^2 = 0 \implies v_{3e} = \sqrt{2 g H}$$

local acceleration:

$$b_l = \frac{\partial v}{\partial t} = \frac{dv_3}{dt} \frac{h_2}{h}, \quad b_l \left(v_3 = \frac{1}{2} v_{3e}, x = \frac{L}{2} \right) = \frac{1}{\bar{L} + L} g H \frac{3}{4} \frac{2 h_2}{h_1 + h_2}$$

convective acceleration:

$$v = v_2 \frac{h_2}{h} \quad \frac{\partial v}{\partial x} = -v_2 h_2 \frac{1}{h^2}$$

$$b_k = v \frac{\partial v}{\partial x} = -v_2^2 \frac{h_2^2}{h^3} \frac{dh}{dx}, \quad \frac{dh}{dx} = \frac{h_2 - h_1}{L}$$

$$b_{k(v_3=\frac{1}{2} v_{3e}, x=\frac{L}{2})} = 4 \frac{g H}{L} \frac{h_2^2 (h_1 - h_2)}{(h_1 + h_2)^3}$$

substantial acceleration:

$$b_s = b_l + b_k = \frac{3}{2} \frac{g H}{\bar{L} + L} \frac{h_2}{h_1 + h_2} + 4 \frac{g H}{L} \frac{h_2^2 (h_1 - h_2)}{(h_1 + h_2)^3}$$